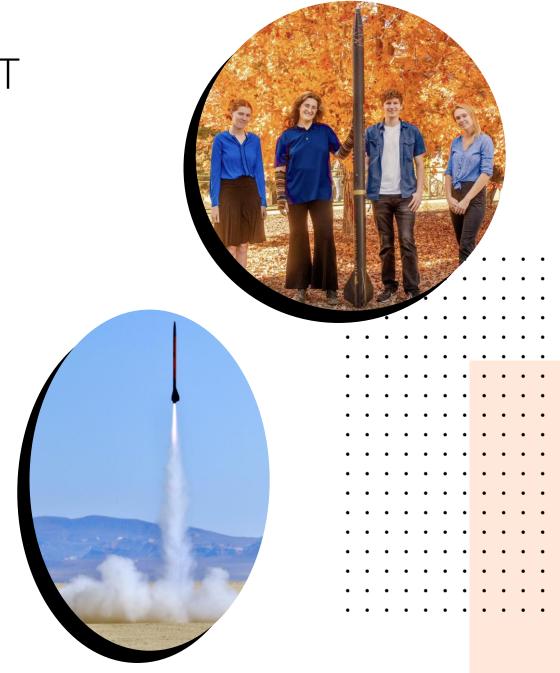
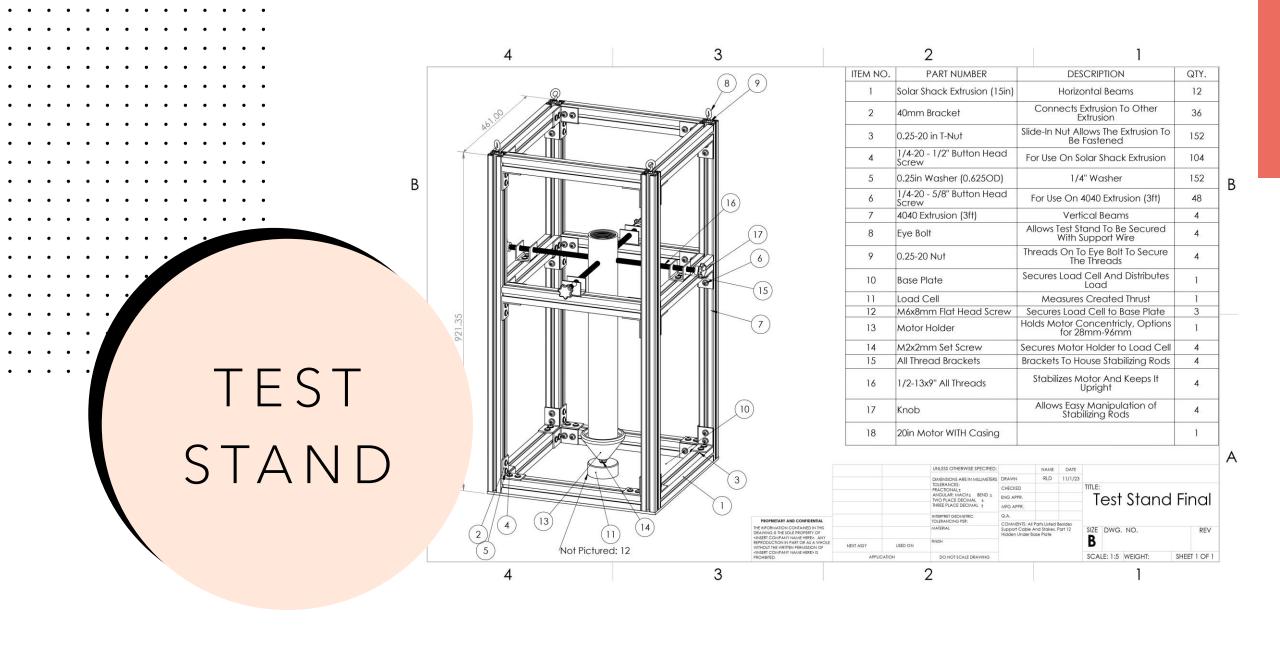


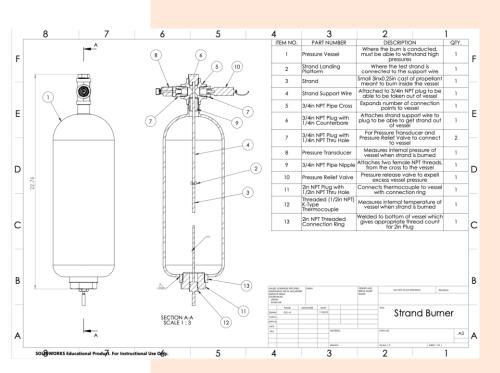
#### SUMMARY OF PROJECT

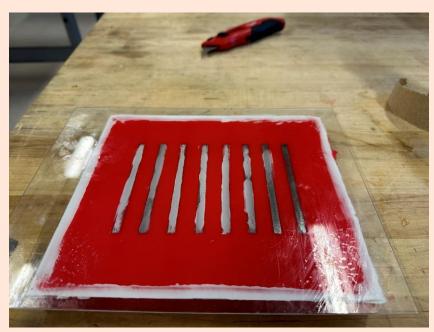
- Build a propulsion system for a high-power, level 2 rocket
  - Design the motor to fit the 3-meter-tall carbon fiber rocket body built by the NAU Rocket Club
  - Develop a unique Ammonium Perchlorate Composite Propellant (APCP) formula
  - Cannot exceed an M-Class motor (>2180 N\*s Impulse)
- Design and build a rocket test stand to gather data on rocket motors efficiency
  - Work with EE team to input thrust and impulse data
- Have at least two motor testing's small scale 38 and 54 mm
- Design and build a motor casing for final 75 mm
- Build a final 75mm diameter rocket motor to launch in Phoenix March 23<sup>rd</sup> or 24<sup>th</sup>, 2024





#### PRESSURE VESSEL STRAND BURNER







# STRAND BURNER TESTING

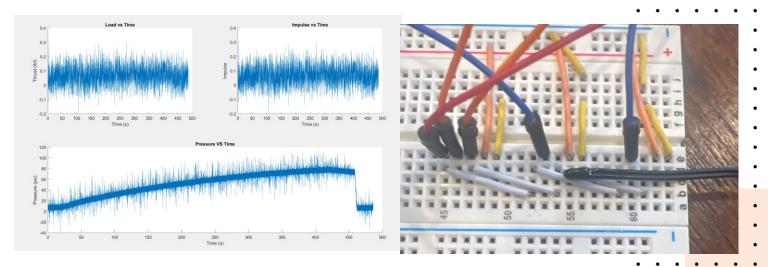






# STRAND BURNER TESTING







#### PROPELLANT FORMULAS

**Batch Sheet** 

Rocket Capstone Batch V2.2

Makes: 2 Strands in the custom mold, 0.25 in x 4 in

Chemical	Amount Planed(g)	Amount Used (g)
R45	4.62	
DOA	0.52	
Silicon Liquid	2 drop	
Aluminum	2.6	
Lamp Black	0.06	
AP Large	8.75	
AP Small	8.75	
MDI Curative	0.68	
Total	25.98	

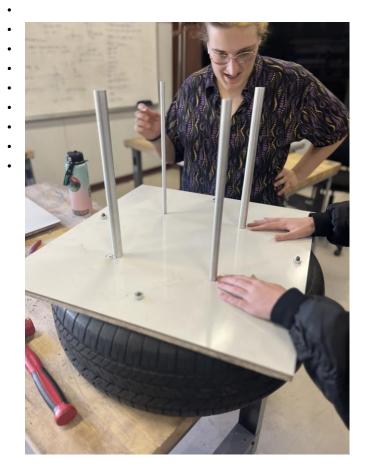
Notes:

Made by adjusting the rations in the modified V2 batch sheet -the Silicon liquid amount is guessed -AP amount is divided equally between large and small -keep all else constant adjust the ratio of AP to AL -2.25% DOA of total mass based on research paper in link below -87.1% for remaining mass is R45, 0% of remaining mass is DOA, 12.9% of remaining mass as in V1

https://pubs.aip.org/aip/acp/article-abstract/2366/1/040009/675922/Preliminary-results-of-DOA-plasticizer-effect-on?redirectedFrom=fulltext

https://www.rocketmotorparts.com/details/p1577809\_7835874.aspx

# PROPELLANT MOTOR CASTING







# NOZZLE MANUFACTURE







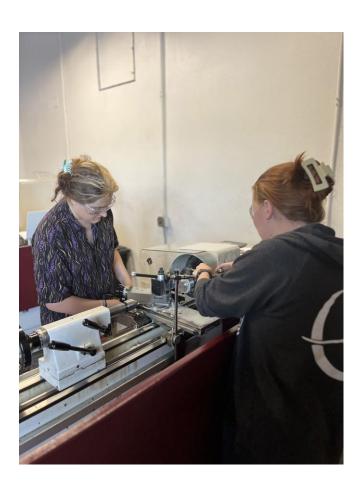
# NOZZLE MANUFACTURE

Ø.754 — Ø.562 —



# BULK HEADS

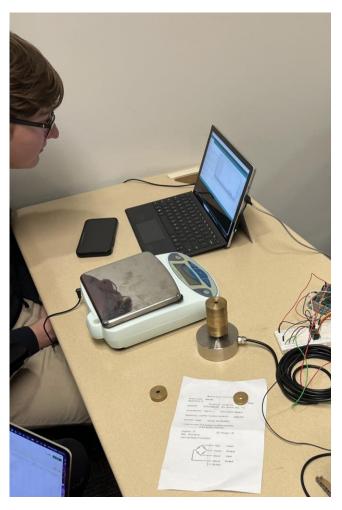


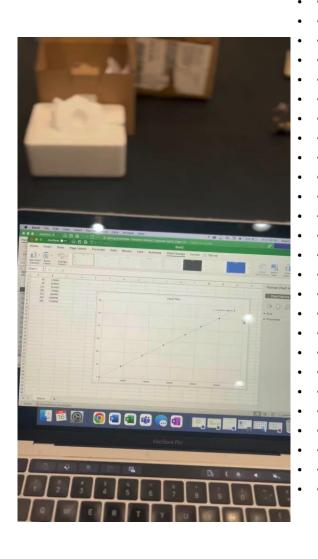




# LOAD CELL AND CALIBRATION







# GANTT CHART AND BOM