

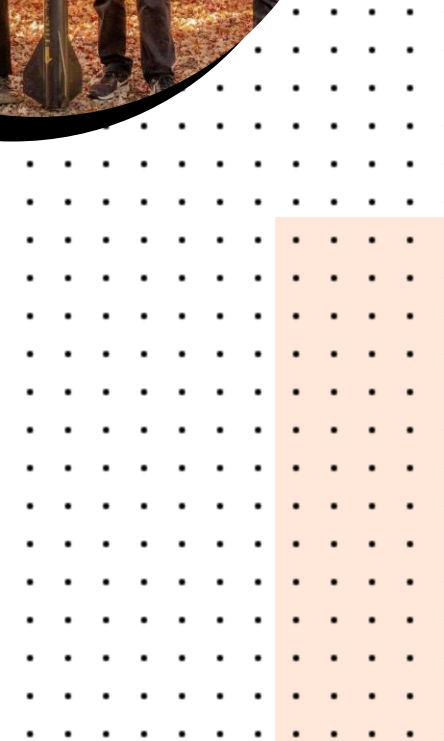
33% BUILD
UPDATE

NAU ROCKET
PROPULSION
TEAM #3

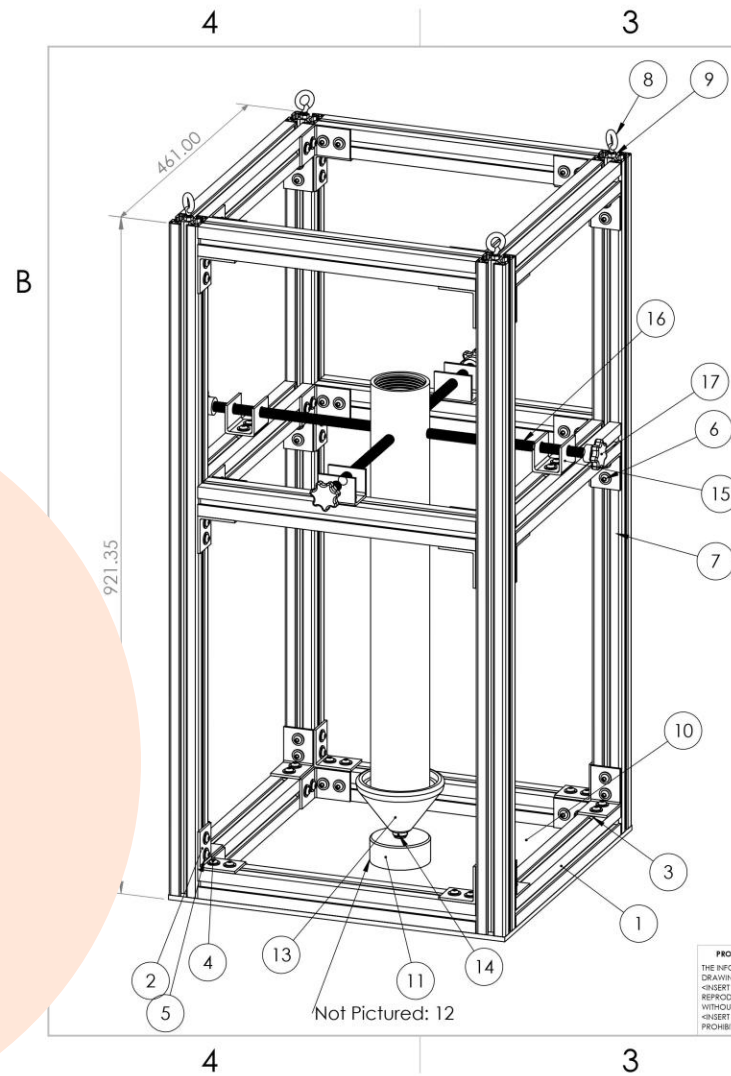
**REMY DASHER, SHANNON COMSTOCK,
GRACE MORRIS AND ANDREW KING**

SUMMARY OF PROJECT

- Build a propulsion system for a high-power, level 2 rocket
 - Design the motor to fit the 3-meter-tall carbon fiber rocket body built by the NAU Rocket Club
 - Develop a unique Ammonium Perchlorate Composite Propellant (APCP) formula
 - Cannot exceed an M-Class motor ($>2180 \text{ N}\cdot\text{s}$ Impulse)
- Design and build a rocket test stand to gather data on rocket motors efficiency
 - Work with EE team to input thrust and impulse data
- Have at least two motor testing's small scale 38 and 54 mm
- Design and build a motor casing for final 75 mm
- Build a final 75mm diameter rocket motor to launch in Phoenix March 23rd or 24th, 2024



TEST STAND



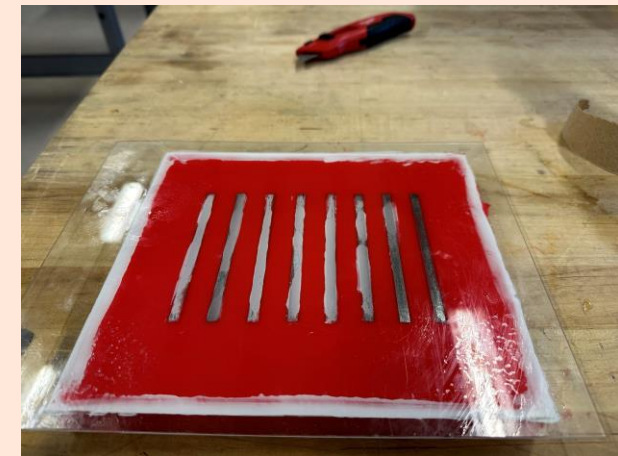
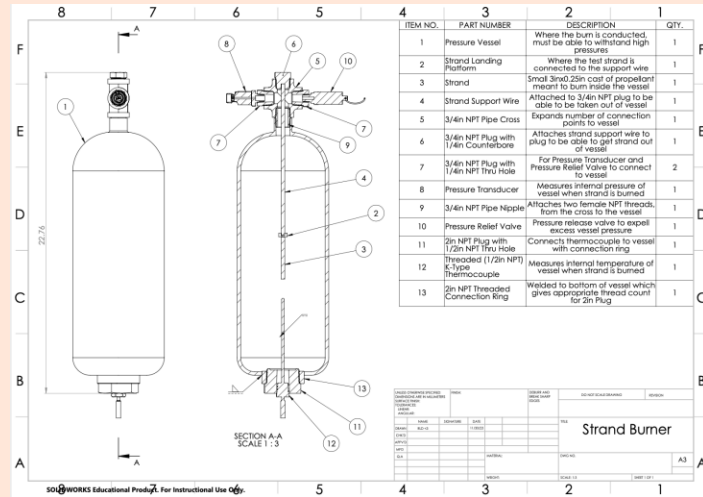
ITEM NO.	PART NUMBER	DESCRIPTION	QTY.
1	Solar Shack Extrusion (15in)	Horizontal Beams	12
2	40mm Bracket	Connects Extrusion To Other Extrusion	36
3	0.25-20 in T-Nut	Slide-In Nut Allows The Extrusion To Be Fastened	152
4	1/4-20 - 1/2" Button Head Screw	For Use On Solar Shack Extrusion	104
5	0.25in Washer (0.625OD)	1/4" Washer	152
6	1/4-20 - 5/8" Button Head Screw	For Use On 4040 Extrusion (3ft)	48
7	4040 Extrusion (3ft)	Vertical Beams	4
8	Eye Bolt	Allows Test Stand To Be Secured With Support Wire	4
9	0.25-20 Nut	Threads On To Eye Bolt To Secure The Threads	4
10	Base Plate	Secures Load Cell And Distributes Load	1
11	Load Cell	Measures Created Thrust	1
12	M6x8mm Flat Head Screw	Secures Load Cell to Base Plate	3
13	Motor Holder	Holds Motor Concentricly, Options for 28mm-96mm	1
14	M2x2mm Set Screw	Secures Motor Holder to Load Cell	4
15	All Thread Brackets	Brackets To House Stabilizing Rods	4
16	1/2-13x9" All Threads	Stabilizes Motor And Keeps It Upright	4
17	Knob	Allows Easy Manipulation of Stabilizing Rods	4
18	20in Motor WITH Casing		1

PROPRIETARY AND CONFIDENTIAL
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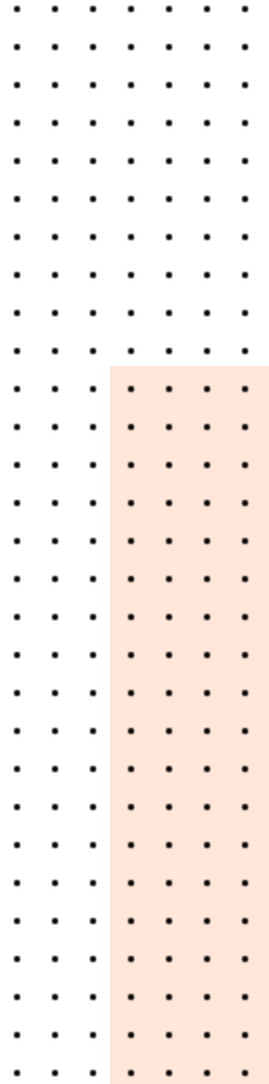
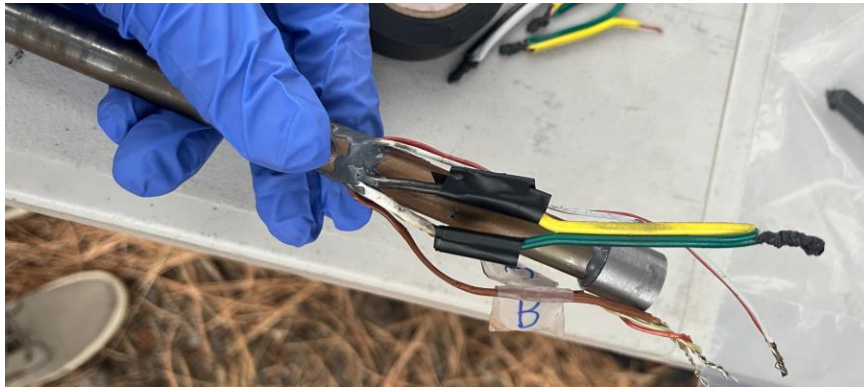
UNLESS OTHERWISE SPECIFIED:		NAME	DATE
DIMENSIONS ARE IN MILLIMETERS		RD	11/1/23
TOLERANCES:		DRAWN	
FRACTIONALS		CHECKED	
ANGULAR: MACH ± BEND ±		ENG APPR.	
TWO PLACE DECIMAL ±		MFG APPR.	
THREE PLACE DECIMAL ±			
INTERPRET GEOMETRIC TOLERANCING PER:		Q.A.	
MATERIAL		COMMENTS: All Parts Listed Besides Support Cable And Stakes. Part 12 Hidden Under Base Plate	
NEXT ASSY	USED ON	FINISH	
APPLICATION		DO NOT SCALE DRAWING	

TITLE:		Test Stand Final	
SIZE	DWG. NO.	REV	
B			
SCALE: 1:5	WEIGHT:	SHEET 1 OF 1	

PRESSURE VESSEL STRAND BURNER

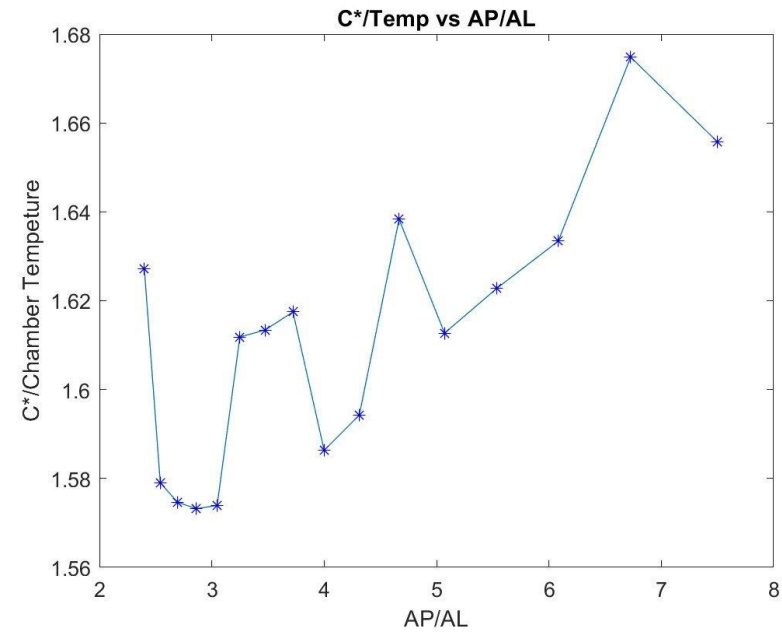


STRAND BURNER TESTING



PROPELLANT FORMULAS

- 74% AP
- 11% AL
- $C^* = 5134.267$ ft/s
- Chamber Temperature = 3065.608 K
- $AP/AL = 6.7273$



P R O P E L L A N T F O R M U L A S

Batch Sheet

Rocket Capstone Batch V2.2

Makes: 2 Strands in the custom mold, 0.25 in x 4 in

Chemical	Amount Planed(g)	Amount Used (g)
R45	4.62	
DOA	0.52	
Silicon Liquid	2 drop	
Aluminum	2.6	
Lamp Black	0.06	
AP Large	8.75	
AP Small	8.75	
MDI Curative	0.68	
Total	25.98	

Notes:

Made by adjusting the rations in the modified V2 batch sheet -the Silicon liquid amount is guessed -AP amount is divided equally between large and small -keep all else constant adjust the ratio of AP to AL -2.25% DOA of total mass based on research paper in link below -87.1% for remaining mass is R45, 0% of remaining mass is DOA, 12.9% of remaining mass as in V1

<https://pubs.aip.org/aip/acp/article-abstract/2366/1/040009/675922/Preliminary-results-of-DOA-plasticizer-effect-on?redirectedFrom=fulltext>

https://www.rocketmotorparts.com/details/p1577809_7835874.aspx

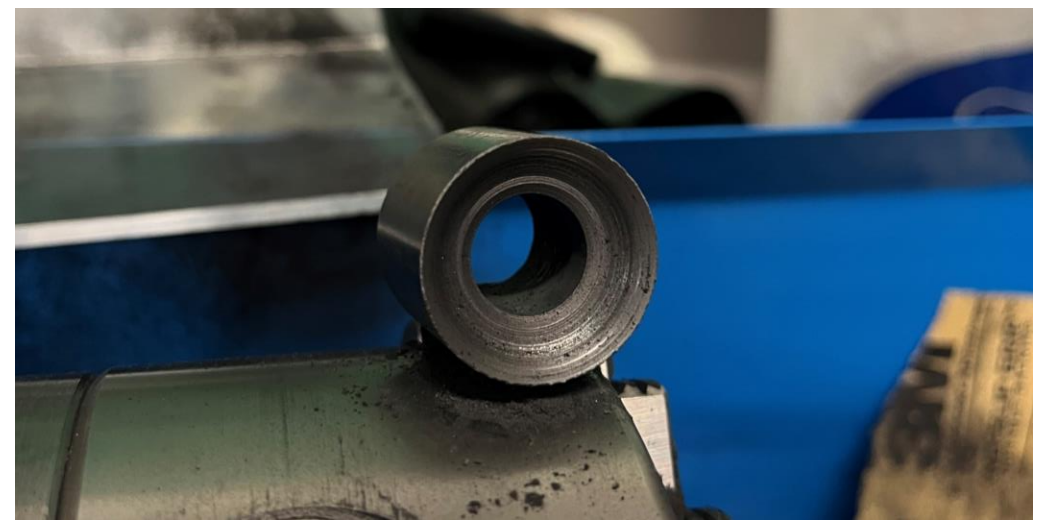
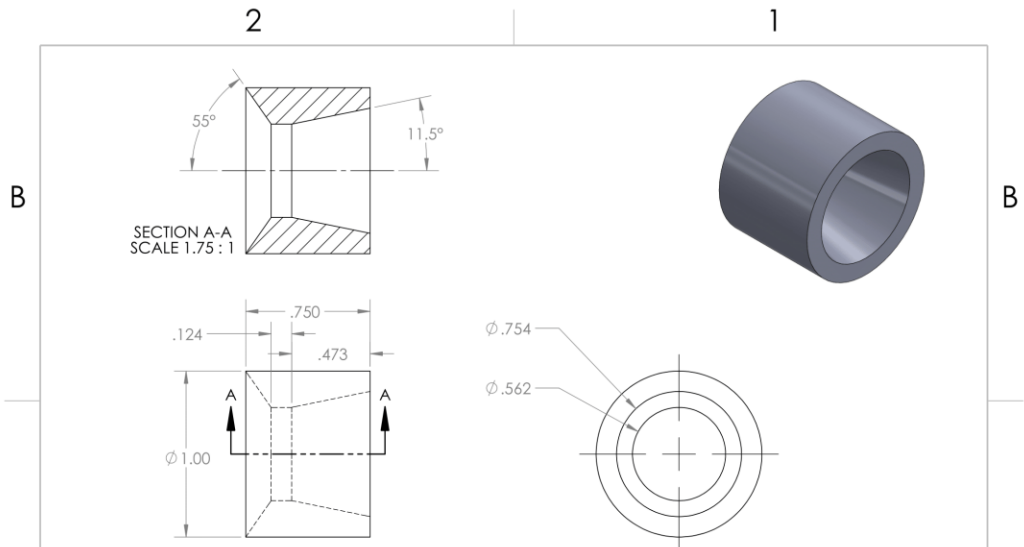
PROPELLANT FORMULATION



NOZZLE MANUFACTURE



NOZZLE MANUFACTURE



MOTOR CASING PURCHASED PARTS

